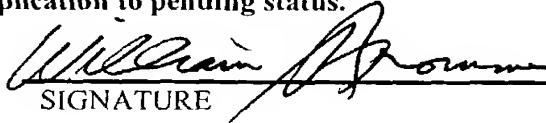


1401891547 JUP 010022
JC02 Rec'd PCT/PTO 03 APR 2002

FORM PTO-1390 (REV 12-2001)		U.S. DEPARTMENT OF COMMERCE PATENT AND TRADEMARK OFFICE		ATTORNEY'S DOCKET NUMBER 452010-2380	
TRANSMITTAL LETTER TO THE UNITED STATES DESIGNATED/ELECTED OFFICE (DO/EO/US) CONCERNING A FILING UNDER 35 U.S.C. 371				U.S. APPLICATION NO. 10/089747 (If known, see 37 CFR 1.55)	
INTERNATIONAL APPLICATION NO. PCT/ZA00/00181		INTERNATIONAL FILING DATE 5 OCTOBER 2000		PRIORITY DATE CLAIMED 5 OCTOBER 1999	
TITLE OF INVENTION MEASURING TWIST DUE TO TORQUE IN A ROTATING SHAFT					
APPLICANT(S) FOR DO/EO/US Bartolomomeus Johannes Le Roux CILLIERS and Roelof Grove BLIGNAUT					
Applicants herewith submits to the United States Designated/Elected Office (DO/EO/US) the following items and other information:					
<ol style="list-style-type: none"> 1. <input checked="" type="checkbox"/> This is a FIRST submission of items concerning a filing under 35 U.S.C. 371. 2. <input type="checkbox"/> This is a SECOND or SUBSEQUENT submission of items concerning a filing under 35 U.S.C. 371 3. <input type="checkbox"/> This is an express request to promptly begin national examination procedures (35 U.S.C. 371(f)) The submission must include items (5), (6), (9) and (21) indicated below. 4. <input checked="" type="checkbox"/> The US has been elected by the expiration of 19 months from the priority date (PCT Article 31). 5. <input checked="" type="checkbox"/> A copy of the International Application as filed (35 U.S.C. 371(c)(2)) <ol style="list-style-type: none"> a. <input type="checkbox"/> is attached hereto (required only if not communicated by the International Bureau). b. <input checked="" type="checkbox"/> has been communicated by the International Bureau. c. <input type="checkbox"/> is not required, as the application was filed in the United States Receiving Office (RO/US). 6. <input checked="" type="checkbox"/> An English language translation of the International Application as filed (35 U.S.C. 371(c)(2)). <ol style="list-style-type: none"> a. <input checked="" type="checkbox"/> is attached hereto. b. <input type="checkbox"/> has been previously submitted under 35 U.S.C. 154(d)(4). 7. <input checked="" type="checkbox"/> Amendments to the claims of the International Application under PCT Article 19 (35 U.S.C. 371(c)(3)) <ol style="list-style-type: none"> a. <input type="checkbox"/> are attached hereto (required only if not communicated by the International Bureau). b. <input type="checkbox"/> have been communicated by the International Bureau. c. <input type="checkbox"/> have not been made; however, the time limit for making such amendments has NOT expired. d. <input checked="" type="checkbox"/> have not been made and will not be made. 8. <input type="checkbox"/> A English language translation of the amendments to the claims under PCT Article 19 (35 U.S.C. 371(c)(3)). 9. <input type="checkbox"/> An oath or declaration of the inventor(s) (35 U.S.C. 371(c)(4)). 10. <input type="checkbox"/> An English language translation of the annexes to the International Preliminary Examination Report under PCT Article 36 (35 U.S.C. 371(c)(5)). 					
Items 11 to 20 below concern document(s) or information included:					
<ol style="list-style-type: none"> 11. <input checked="" type="checkbox"/> An Information Disclosure Statement under 37 CFR 1.97 and 1.98. 12. <input type="checkbox"/> An assignment document for recording A separate cover sheet in compliance with 37 CFR 3.28 and 3.31 is included 13. <input checked="" type="checkbox"/> A FIRST preliminary amendment. 14. <input type="checkbox"/> A SECOND or SUBSEQUENT preliminary amendment. 15. <input type="checkbox"/> A substitute specification. 16. <input type="checkbox"/> A change of power of attorney and/or address letter. 17. <input type="checkbox"/> A computer-readable form of the sequence listing in accordance with PCT Rule 13ter.2 and 35 U.S.C. 1.821 - 1.825 18. <input type="checkbox"/> A second copy of the published international application under 35 U.S.C. 154(d)(4). 19. <input type="checkbox"/> A second copy of the English language translation of the international application under 35 U.S.C. 154(d)(4). 20. <input checked="" type="checkbox"/> Other items or information: PCT/IPEA/409, PCT/ISA/210, 2 sheets of drawings, 1 page abstract, Claim of Priority 					
				<p align="center">EXPRESS MAIL</p> <p>Mailing Label Number: EV073647285US</p> <p>Date of Deposit April 3, 2002</p> <p>I hereby certify that this paper or fee is being deposited with the United States Postal Service</p> <p>"Express Mail Post Office to Addressee" Service under 37 CFR 1.10 on the date indicated above and is addressed to the Assistant Commissioner for Patents and Trademarks, Box PCT Washington, DC 20231</p> <p>_____ (Typed or printed name of person mailing paper or fee)</p> <p>_____ (Signature of person mailing paper or fee)</p>	

US APPLICATION NO (If known, see 37 C.F.R. 1.50) 10/089747		INTERNATIONAL APPLICATION NO PCT/ZA00/00181		ATTORNEY'S DOCKET NO 452010-2380	
21. <input checked="" type="checkbox"/> The following fees are submitted:				CALCULATIONS PTO USE ONLY	
BASIC NATIONAL FEE (37 CFR 1.492(a)(1)-(5): Neither international preliminary examination fee (37 CFR 1.482) nor international search fee (37 CFR 1.445(a)(2)) paid to USPTO and International Search Report not prepared by the EPO or JPO \$1040 00 International preliminary examination fee (37 C.F.R. 1.482) not paid to USPTO but International Search Report prepared by the EPO or JPO \$890 00 International preliminary examination fee (37 CFR 1.482) not paid to USPTO but international search fee (37 CFR 1.445(a)(2)) paid to USPTO... .. \$740 00 International preliminary examination fee paid to USPTO (37 CFR 1.482) but all claims did not satisfy provisions of PCT Article 33(1)-(4) \$710 00 International preliminary examination fee paid to USPTO (37 CFR 1.482) and all claims satisfied provisions of PCT Article 33(1)-(4) \$100 00					
ENTER APPROPRIATE BASIC FEE AMOUNT =				\$ 890.00	
Surcharge of \$130.00 for furnishing the oath or declaration later than <input type="checkbox"/> 20 <input type="checkbox"/> 30 months from the earliest claimed priority date (37 CFR 1.492(e)).				\$	
CLAIMS	NUMBER FILED	NUMBER EXTRA	RATE	\$	
Total Claims	<u>35</u> - 20 =	<u>15</u>	x \$18.00	\$ 270.00	
Independent Claims	<u>3</u> - 3 =	<u>0</u>	x \$84.00	\$ 0.00	
MULTIPLE DEPENDENT CLAIM(S) (if applicable)			+ \$280.00	\$	
TOTAL OF ABOVE CALCULATIONS =				\$	
<input type="checkbox"/> Applicant claims small entity status. See 37 C.F.R. 1.27. The fees indicated above are reduced by 1/2				\$	
SUBTOTAL =				\$1160.00	
Processing fee of \$130.00 for furnishing the English translation later than <input type="checkbox"/> 20 <input type="checkbox"/> 30 months from the earliest claimed priority date (37 CFR 1.492(f))				\$	
TOTAL NATIONAL FEE =				\$	
Fee for recording the enclosed assignments (37 CFR 1.21(h)) The assignment must be accompanied by an appropriate cover sheet (37 CFR 3.28, 3.31) \$40.00 per property				\$	
TOTAL FEES ENCLOSED =				\$1160.00	
				Amount to be refunded:	\$
				charged:	\$
a. <input checked="" type="checkbox"/> A check in the amount of \$1160.00 to cover the above fees is enclosed. b. <input type="checkbox"/> Please charge my Deposit Account No. _____ in the amount of \$ _____ to cover the above fees. A duplicate copy of this sheet is enclosed. c. <input checked="" type="checkbox"/> The Commissioner is hereby authorized to charge any additional fees which may be required, or credit any overpayment to Deposit Account No. 50-0320 . A duplicate copy of this sheet is enclosed. d. <input type="checkbox"/> Fees are to be charged to a credit card. WARNING: Information on this form may become public. Credit card information should not be included on this form. Provide credit card information and authorization on PTO-2038.					
NOTE: Where an appropriate time limit under 37 CFR 1.494 or 1.495 has not been met, a petition to revive (37 CFR 1.137(a) or (b)) must be filed and granted to restore the application to pending status.					
SEND ALL CORRESPONDENCE TO: William S. Frommer, Esq. Frommer Lawrence & Haug LLP 745 Fifth Avenue New York, New York 10151					
				 SIGNATURE	
				William S. Frommer NAME	
				25,506 REGISTRATION NUMBER	
Dated: <u>April 3, 2002</u>					

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PATENT
451010-2380 10/089747

JC10 Rec'd PCT/PTO 03 APR 2002

IN THE UNITED STATES PATENT AND TRADEMARK OFFICE

Applicant: CILLIERS, et al.
Filed.: Filed Concurrently Herewith
Title of Invention: MEASURING TWIST DUE TO TORQUE IN A
ROTATING SHAFT

745 Fifth Avenue
New York, NY 10151

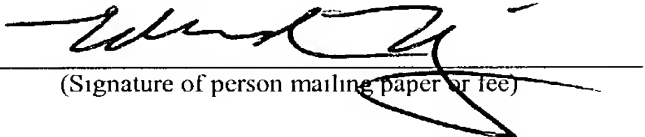
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Date of Deposit: April 3, 2002

I hereby certify that this paper or fee is being deposited with the United States Postal Service "Express Mail Post Office to Addressee" Service under 37 CFR 1.10 on the date indicated above and is addressed to the Assistant Commissioner of Patents and Trademarks, Washington, DC 20231

Edward Nay
(Typed or printed name of person mailing paper or fee)


(Signature of person mailing paper or fee)

PRELIMINARY AMENDMENT

Assistant Commissioner for Patents
Box PCT
Washington, D.C. 20231

Sir:

Before the issuance of the first Office Action, please amend the above-identified application as follows:

IN THE SPECIFICATION:

Please add a new paragraph beginning at page 1, line 7, as follows:

--International Patent Document WO 96/20395 discloses a process and device for determining a power output rate between a rotary shaft and one or more stationary machines

IN THE CLAIMS:

Please add new claims 37-71 as follows:

establishing a datum time period between a datum time moment the datum point passes a stationary datum station and a time moment the measure point passes a stationary measure station under a no load condition and recording rotational speed of the shaft as the datum speed;

calculating the twist in the shaft on the basis of the difference between the measure time period and the datum time period, the measure speed and the length over which torque is applied.

39. A method as claimed in Claim 38 in which the shaft is a crankshaft of a reciprocating internal combustion engine.

40. A method as claimed in Claim 39 in which the reciprocating engine includes a ring gear having gear teeth at one end of the crankshaft and a disc at an opposed end of the crankshaft, the datum point being on the disc, the measure point being on a gear tooth of the ring gear.

41. A method as claimed in Claim 40 which is performed in respect of a plurality of measure points and a plurality of datum points, the plurality of measure points being on a corresponding plurality of gear teeth of the ring gear and the plurality of datum points being points which are circumferentially spaced on the disc.

42. A method as claimed in Claim 41 in which the reciprocating internal combustion engine includes a plurality of cylinders, the method being performed in respect of each cylinder.

43. A method as claimed in Claim 41 which is performed for each power stroke of the reciprocating internal combustion engine.

44. A method as claimed in Claim 38 in which the shaft is a main shaft of a gas turbine engine.

45. A method as claimed in Claim 44 in which at least one of the datum point and the measure point is on a vane of at least one of a compressor and a turbine of the gas turbine engine:

46. A method as claimed in Claim 45 which is preformed in respect of a plurality of measure points and a plurality of datum points, the plurality of measure points being on a corresponding plurality of vanes of one of the compressor and the turbine and the plurality of datum points being on a corresponding plurality of vanes of the other of the compressor and the turbine, the sensors being external of casings surrounding respectively the compressor and the turbine.

47. A method as claimed in Claim 37 in which establishing the time moments includes triggering a sensor at respectively the datum station and the measure station by means

of triggers at correspondingly the or each datum point and the or each measure point, creating a signal by each sensor and recording the signal against time.

48. A method as claimed in Claim 47 in which the triggers are masses of magnetic material, and the sensors are responsive to said magnetic material to create said signals.

49. A method as claimed in Claim 47 in which the triggers are optically detectable surfaces, and the sensors sense passing of the triggers. optically.

50. A method of measuring twist in a rotating shaft which is subjected to torque including sensing and recording a datum time moment when a datum point on the shaft passes a fixed datum station; sensing, at a fixed measure station, an arrival time moment of a measure point on the shaft, longitudinally spaced from said datum point by a predetermined distance; measuring a measure time period between said datum time moment and said arrival time moment; measuring rotational speed of the shaft; comparing the measure time period with a computed time period under a no load condition; and calculating the twist in the shaft on the basis of the time lag and the rotational speed.

51. A method as claimed in Claim 50 which includes empirically predetermining a datum time period between a datum time moment and an arrival time moment under a no load condition at a determined rational speed, and computing a computed time period and a computed arrival moment for any specific rotational speed on the basis of the datum time period bearing in mind the respective rotational speeds.

52. A measuring apparatus for measuring twist in a rotating shaft which is subjected to torque, the measuring apparatus including

at least one datum trigger at a datum point on the shaft;

a datum sensor at a stationary datum station arranged to sense said at least one datum trigger when said at least one datum point is in register with the datum station and to generate correspondingly at least one datum signal;

at least one measure trigger at correspondingly at least one measure point on the shaft longitudinally spaced from said at least one datum point;

a measure sensor at a stationary measure station arranged to sense said at least one measure trigger when said at least one measure trigger is in register with the measure station and to generate correspondingly at least one measure signal;

a clock keeping time;

a rotational speed meter for measuring and recording rotational speeds of the shaft against time respectively as the datum speed and as the measure speed;

recording means for recording said datum and measure signals against time;

a processor programmed to establish

correspondingly at least one datum time period between a time moment said at least one datum point passes the datum station and a time moment said at least one measure point passes the measure station under no load conditions and recording the rotational speed of the shaft as the datum speed,

correspondingly at least one measure time period between a time moment said at least one datum point passes the stationary datum station and a time moment said at least one measure point passes the stationary measure station when the shaft is subjected to torque and recording the rotational speed of the shaft as the measure speed, and

the twist in the shaft on the basis of the difference between said at least one measure time period and said at least one datum time period, the measure speed and a length of the shaft over which torque is applied.

53. A measuring apparatus as claimed in Claim 52 in which the shaft includes a torque point at which torque is applied to the shaft in use and a load point at which a load is connected to the shaft in use, the torque point and the load point being longitudinally spaced, the shaft, in use, being stressed and undergoing twist between the torque point and the load point, one of the or each measure point and the or each datum point being positioned along said stressed portion of the shaft, the other of the or each measure point and the or each datum point being positioned in a relaxed portion of the shaft beyond one of the torque point and the load point.

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54. A measuring apparatus as claimed in Claim 53 in which the shaft is a crankshaft of a reciprocating internal combustion engine.
55. A measuring apparatus as claimed in Claim 54 which includes a plurality of measure points and a plurality of datum points.
56. A measuring apparatus as claimed in Claim 55 in which the reciprocating engine includes a ring gear having gear teeth at one end of the crankshaft and a disc at an opposed end of the crankshaft, the datum points and the measuring points being respectively on gear teeth of the ring gear and at circumferentially spaced positions on the disc.
57. A measuring apparatus as claimed in Claim 53 in which the shaft is a main shaft of a gas turbine engine.
58. A measuring apparatus as claimed in Claim 57 in which at least one of the datum point and the measure point is on correspondingly at least one of a vane of a compressor and a vane of a turbine of the gas turbine engine.
59. A measuring apparatus as claimed in Claim 58 which includes a plurality of datum points and a plurality of measure points, the datum points and the measure points being respectively on vanes of the compressor and the turbine.
60. A measuring apparatus as claimed in Claim 52 in which the triggers are masses of magnetic material, and the sensors are responsive to said magnetic material to generate signals.
61. A measuring apparatus as claimed in Claim 60 in which the sensors are in the form of magnetic sensors arranged external of casings surrounding respectively the compressor and the turbine.

62. A measuring apparatus as claimed Claim 52 in which the triggers are optically detectable surfaces, and the sensors are optical sensors responsive to the optically detectable triggers to generate signals.

63. A method of measuring torque in a rotating shaft, including measuring twist in the rotating shaft in accordance with Claim 37, and calculating torque on the basis of the measured twist bearing in mind a pre-established length of the shaft over which length said twist is measured, and pre-established physical characteristics of the shaft determining torque-twist behaviour of the shaft.

64. A method as claimed in Claim 63 in which said physical characteristics of the shaft determining torque-twist behaviour of the shaft are established empirically.

65. A measuring apparatus for measuring torque in a rotating shaft, which includes a measuring apparatus for measuring twist as claimed in Claim 52, in which the processor is programmed to calculate torque on the basis of the measured twist bearing in mind a pre-established length of the shaft over which length said twist is measured and pre-established physical characteristics of the shaft determining torque-twist behaviour of the shaft.

66. A method of managing operation of an engine including measuring a quantity related to torque in a drive shaft of the engine as claimed in Claim 37, comparing the measured value of said quantity related to torque to a predetermined standard value of said quantity, establishing any deviation between the measured value and the standard value and controlling an operating function of the engine in response to said establish deviation.

67. A method as claimed in Claim 66 in which said operating function of the engine is at least one of ignition timing when the engine is a spark ignition engine, injector pump timing when said engine is a diesel engine, boost pressure when said engine is turbo charged or supercharged.

68. A method as claimed in Claim 66 including automatically calibrating the measuring apparatus in respect of datum time periods when the engine is operated under a no load condition.

69. An engine including
a measuring apparatus as claimed in Claim 52;
an engine management apparatus including a comparator arranged to receive a signal indicative of a measured value of a quantity related to torque in a drive shaft of the engine from the measuring apparatus, the comparator being preprogrammed to compare said measured value with a standard value of said quantity related to torque to generate a control signal, the engine management apparatus being responsive to said control signal to control an operating function of the engine.

70. An engine as claimed in Claim 69 in which said operating function of the engine is at least one of ignition timing when the engine is a spark ignition engine, injector pump timing when said engine is a diesel engine, boost pressure when said engine is turbocharged or supercharged.

71. An engine as claimed in Claim 70 in which the engine management apparatus is preprogrammed to calibrate the measuring apparatus in respect of datum time periods when the engine is operated under a no load condition.

MEASURING TWIST DUE TO TORQUE IN A ROTATING SHAFT

THIS INVENTION relates to measuring twist due to torque in a rotating shaft. It relates more specifically to a method of and measuring apparatus for measuring twist in a rotating shaft subjected to torque, to a method of and a measuring apparatus for measuring torque in a rotating shaft, and to a method of and an apparatus for managing operation of an engine.

International Patent Document WO 96/20395 discloses a process and device for determining a power output rate between a rotary shaft and one or more stationary machines connected to the shaft. The shaft is subjected to torque causing torsional deformation which is measured as an angle. Power output is determined as the product of the angle and a predetermined calibration coefficient.

The term "twist" is for convenience used for purposes of this invention to denote angular deflection or angular deformation.

In accordance with a first aspect of this invention, broadly, there is provided a method of measuring twist in a rotating shaft which is subjected to torque including sensing an arrival time moment of a measure point on the shaft at a fixed measure station, comparing the arrival time moment with a computed arrival time moment of said measure point at said measure station under a no load condition, measuring rotational speed of the shaft, and calculating the twist in the shaft on the basis of the time lag and the rotational speed.

The method may include sensing and recording a datum time moment when a datum point on the shaft, longitudinally spaced from said measure point by a predetermined distance, passes a fixed datum station, and measuring a measure time period between said datum time moment and said arrival time moment.

The method may include empirically predetermining a datum time period between a datum time moment and an arrival time moment under a no load condition at a determined rotational speed, and computing a computed time period and a computed arrival time moment for any specific rotational speed on the basis

point passes a stationary datum station and a time moment the measure point passes a stationary measure station under a no load condition and recording rotational speed of the shaft as the datum speed;

5 measuring a measure time period between a time moment the datum point passes the stationary datum station and a time moment the measure point passes the stationary measure station when the shaft is subjected to torque, recording the rotational speed of the shaft as the measure speed and establishing longitudinal positions respectively at which torque is applied and of the datum point and of the measure point;

10 calculating the twist in the shaft from the difference between the measure time period and the datum time period, bearing in mind the difference between the measure speed and the datum speed.

A torque point at which torque is applied to the shaft and a load point at which a load is connected to the shaft may be longitudinally spaced, the shaft being stressed and undergoing twist between the torque point and the load point, one of the measure point and the datum point being positioned along said stressed portion of the shaft, the other of the measure point and the datum point being positioned in a relaxed portion of the shaft beyond one of the torque point and the load point.

20 Advantageously, the method may be applied to a shaft in the form of a crankshaft of a reciprocating internal combustion engine. The reciprocating engine may include a ring gear having gear teeth at one end of the crankshaft and a disc at an opposed end of the crankshaft, the datum point being on the disc, the measure point being on a gear tooth of the ring gear. The method is preferably
25 performed in respect of a plurality of measure points and a plurality of datum points, the plurality of measure points being on a corresponding plurality of gear teeth of the ring gear and the plurality of datum points being points which are circumferentially spaced on the disc. Advantageously, the number of datum points and the number of measure points may be equal, to allow the respective
30 datum points and measure points to be associated on a one-on-one basis.

When the reciprocating internal combustion engine includes a plurality of cylinders, the method may be performed in respect of each cylinder. Furthermore, the method may be performed for each power stroke of the

reciprocating internal combustion engine.

In another technical application, the method may be applied to a gas turbine engine, the shaft then being a main shaft of the gas turbine engine.

Then, at least one of the datum point and the measure point may be
5 on a vane of at least one of a compressor and a turbine of the gas turbine engine.
However, the method is preferably performed in respect of a plurality of measure points and a plurality of datum points, the plurality of measure points being on a corresponding plurality of vanes of one of the compressor and the turbine and the plurality of datum points being on a corresponding plurality of vanes of the other
10 of the compressor and the turbine, the sensors being external of casings surrounding respectively the compressor and the turbine.

Generally, establishing the time moments may include triggering a sensor at respectively the datum station and the measure station by means of triggers at correspondingly the or each datum point and the or each measure
15 point, creating a signal by each sensor and recording the signal against time.

The triggers may be masses of magnetic material, and the sensors may be responsive to said magnetic material to create said signals.

Instead, the triggers may be optically detectable surfaces, the sensors then sensing passing of the triggers optically.

20 In accordance with a second aspect of this invention, there is provided measuring apparatus for measuring twist in a rotating shaft which is subjected to torque, the measuring apparatus including
at least one datum trigger at a datum point on the shaft;
a datum sensor at a stationary datum station arranged to sense said at least
25 one datum trigger when said at least one datum point is in register with the datum station and to generate correspondingly at least one datum signal;
at least one measure trigger at correspondingly at least one measure point on the shaft longitudinally spaced from said at least one datum point;
a measure sensor at a stationary measure station arranged to sense said at
30 least one measure trigger when said at least one measure trigger is in register with

the measure station and to generate correspondingly at least one measure signal;
a clock keeping time;

a rotational speed meter for measuring and recording rotational speeds of
the shaft against time respectively as the datum speed and as the measure speed;

5 recording means for recording said datum and measure signals against time;
a processor programmed to establish

correspondingly at least one datum time period between a time
moment said at least one datum point passes the datum station and a time
moment said at least one measure point passes the measure station under no load
10 conditions and recording the rotational speed of the shaft as the datum speed,

correspondingly at least one measure time period between a time
moment said at least one datum point passes the stationary datum station and a
time moment said at least one measure point passes the stationary measure
station when the shaft is subjected to torque and recording the rotational speed
15 of the shaft as the measure speed, and

the twist of the shaft from the difference between said at least one
measure time period and said at least one datum time period, bearing in mind the
ratio between the measure speed and the datum speed.

The shaft may include a torque point at which torque is applied to
20 the shaft in use and a load point at which a load is connected to the shaft in use,
the torque point and the load point being longitudinally spaced, the shaft, in use,
being stressed and undergoing twist between the torque point and the load point,
one of the or each measure point and the or each datum point being positioned
along said stressed portion of the shaft, the other of the or each measure point
25 and the or each datum point being positioned in a relaxed portion of the shaft
beyond one of the torque point and the load point.

The shaft may advantageously be a crankshaft of a reciprocating
internal combustion engine. Then the measuring apparatus may preferably include
a plurality of measure points and a plurality of datum points.

30 Advantageously, the reciprocating engine may include a ring gear
having gear teeth at one end of the crankshaft and a disc at an opposed end of
the crankshaft, the datum points and the measuring points being respectively on
gear teeth of the ring gear and at circumferentially spaced positions on the disc.

In another technical application, the shaft may be a main shaft of a gas turbine engine. Then, at least one of the datum point and the measure point is on correspondingly at least one of a vane of a compressor and a vane of a turbine of the gas turbine engine. However, the measuring apparatus may include
5 a plurality of datum points and a plurality of measure points, the datum points and the measure points being respectively on vanes of the compressor and the turbine.

The triggers may be masses of magnetic material, and the sensors may be magnetic sensors responsive to said magnetic material to generate signals. The magnetic sensors may be arranged ~~externally~~ of casings surrounding
10 respectively the compressor and the turbine.

Instead, the triggers may be optically detectable surfaces, and the sensors may be optical sensors responsive to the optically detectable triggers to generate signals.

In accordance with a third aspect of this invention, there is provided
15 a method of measuring torque in a rotating shaft, including measuring twist in the rotating shaft in accordance with the first aspect of this invention, and calculating torque on the basis of the measured twist bearing in mind a pre-established length of the shaft over which length said twist is measured, and pre-established physical characteristics of the shaft determining torque-twist behaviour of the shaft.

20 Said physical characteristics of the shaft determining torque-twist behaviour of the shaft may advantageously be established empirically.

In accordance with a fourth aspect of this invention, there is provided a measuring apparatus for measuring torque in a rotating shaft, which includes a measuring apparatus for measuring twist in accordance with the second
25 aspect of this invention, in which the processor is programmed to calculate torque on the basis of the measured twist bearing in mind a pre-established length of the shaft over which length said twist is measured and pre-established physical characteristics of the shaft determining torque-twist behaviour of the shaft.

In accordance with a fifth aspect of this invention, there is provided
30 a method of managing operation of an engine including measuring a quantity

related to torque in a drive shaft of the engine in accordance with the first aspect or the third aspect of this invention, comparing the measured value of said quantity related to torque to a predetermined standard value of said quantity, establishing any deviation between the measured value and the standard value and
5 controlling an operating function of the engine in response to said establish deviation.

Said operating function of the engine may be at least one of ignition timing when the engine is a spark ignition engine, injector pump timing when said engine is a diesel engine, boost pressure when said engine is turbocharged or
10 supercharged.

Advantageously, the method may include automatically calibrating the measuring apparatus in respect of datum time periods when the engine is operated under a no load condition.

In accordance with a sixth aspect of this invention, there is provided
15 an engine including

a measuring apparatus in accordance with the second aspect or the fourth aspect of this invention;

an engine management apparatus including a comparator arranged to receive a signal indicative of a measured value of a quantity related to torque in
20 a drive shaft of the engine from the measuring apparatus, the comparator being preprogrammed to compare said measured value with a standard value of said quantity related to torque to generate a control signal, the engine management apparatus being responsive to said control signal to control an operating function of the engine.

25 Said operating function of the engine may be at least one of ignition timing when the engine is a spark ignition engine, injector pump timing when said engine is a diesel engine, boost pressure when said engine is turbocharged or supercharged.

Advantageously, the engine management apparatus may be
30 preprogrammed to calibrate the measuring apparatus in respect of datum time periods when the engine is operated under a no load condition.

The invention will now be described, by way of examples, with reference to the accompanying diagrammatic drawings.

In the drawings

Figure 1 shows a block diagram of an apparatus for measuring torque in accordance with the invention;

Figure 2 shows, in three dimensional view, the invention applied to a reciprocating internal combustion engine; and

Figure 3 corresponds to Figure 2, but shows a gas turbine engine.

Referring to Figures 1 and 2 of the drawings; reference numeral 10 generally indicates a measuring apparatus in accordance with the invention.

The apparatus 10 measures twist in a drive shaft in the form of a crankshaft 30 (see Figure 2 of the drawings) of a reciprocating internal combustion engine 29. In Figure 2 only the pistons 32 of the engine, and connection of the pistons 32 via connecting rods 34 to cranks of ^{the} crankshaft 30 of the engine, are shown.

A flywheel 40 incorporating a ring gear having a plurality of circumferentially spaced teeth 42 is mounted at one end of the crankshaft 30. The flywheel 40 and specifically also the teeth 42 are of a magnetic material. During operation of the engine, thrust of the pistons 32 during power strokes causes twist in the crankshaft 30.

In order to measure said twist, the measuring apparatus 10 includes a circular metallic disc 50 secured to the end of the crankshaft 30 opposite the flywheel 40. The disc 50 has a plurality of formations or teeth 52 defining triggers in the form of datum points. The teeth, in this embodiment, are of magnetic material.

The apparatus 10 further includes a sensor in the form of a magnetic sensor or pick-up device 12.1, e.g. Part No. 304166 from the RS catalogue, mounted on the engine adjacent a pitch circle of the teeth 52 of the disc so as to be registered sequentially with the teeth as the disc 50 rotates with the crankshaft 30. Each time a tooth 52 is in register with the sensor 12.1 a signal in the form

of a pulse is produced, and is conveyed to the apparatus 10 as shown at 13.1. The teeth define triggers in the form of datum points.

The apparatus 10 also includes a sensor in the form of a magnetic sensor or pick-up device 12.2, e.g. Part No. 304166 from the RS catalogue, mounted on the engine adjacent a pitch circle of the teeth 42 of the ring gear on the fly wheel 40 so as to be registered sequentially with the teeth as the flywheel 40 rotates with the crankshaft 30. Each time a tooth 42 is in register with the sensor 12.2, a signal in the form of a pulse is produced and is conveyed to the apparatus 10 as shown at 13.2. The teeth 42 define triggers in the form of measure points.

In another embodiment, the sensors 12.1, 12.2 may be optical sensors and the triggers 42, 52 may be optically detectable surfaces.

For convenience, the sensors 12.1, 12.2 or the signal outputs 13.1, 13.2 are shown at 12 in Figure 1.

15 As the crank shaft 30 rotates, when each datum point 52 passes the sensor 12.1, a signal pulse is generated and is recorded in a processor 14 (Figure 1) including a clock 18, thus setting a datum time moment. The clock 18 typically operates at a frequency of 10 MHz.

Assume the crank shaft 30 rotates under no load and at a rotational speed which is measured and recorded (conveniently in the processor 14 by means of the signal pulses generated by the sensor 12.1 on being passed by the datum points 52). Being under no load, the crank shaft 30 is under zero twist. The arrival time moments of the respective triggers 52 are sensed and are recorded as are the respective time periods between respective datum time moments and the respective arrival time moments - bearing in mind the rotational speed of the crank shaft 30. From this empirically derived data, computed arrival time periods, under no load conditions, can be computed by the processor 14 for any given or measured rotational speed. Advantageously, the number of datum points and the number of measure points are equal and they can then advantageously be associated on a one-on-one basis. If the number of points are not the same, each measure point is associated with an appropriate or most

appropriate datum point, for example on the basis of minimizing the time periods.

When the crank shaft 30 rotates under load, the power strokes of the pistons 32 impart torque pulses to the crank shaft causing twist in the crank shaft. The amount of twist depends on physical characteristics of the crank shaft, e.g. effective moment of inertia and the torsional Young's modulus of the material of the crank shaft, the effective length over which the torque is applied (i.e. the longitudinal distance between the respective big end bearing and the flywheel 40), and the magnitude of the torque. All of the above except the magnitude of the torque remain constant for each piston 32, and can be pre-established for each engine or type of engine, advantageously empirically. Thus, the amount of twist uniquely defines the magnitude of the torque.

Furthermore, due to the twist, the respective triggers 42 arrive at the sensor 12.2 late relative to a respective computed arrival time moment. The amount of time lag, bearing in mind the rotational speed and the length over which the torque is transmitted, uniquely defines the twist.

The processor 14 is programmed to measure the respective time lags and to calculate the respective twists, as mentioned above, taking into account the rotational speed and the lengths over which the torque is transmitted, for the respective piston power strokes which are obtained from the firing order of the specific engine. From the calculated twist, the torque applied in each power stroke of each piston is calculated.

The apparatus 10 accordingly includes the processor 14 including a comparator 16 into which the respective signals are input from item 12.

Arrival time moments are compared by the comparator 16, from which the time periods are measured by the clock 18. Typically, the clock 18 operates at a frequency of 10 MHz, as mentioned above. The Applicants expect such resolution to be sufficient for purposes of this invention for use in commercial reciprocation engines.

The time periods are compared and twist, and ultimately torque values, are calculated as explained above. Results are stored in a data storage

means 20, overwriting any previously stored value therein every second revolution in the case of a four-stroke engine.

The processor 14 further includes a second comparator 21 which includes a dataset prepared before installation and characterizing the engine when it is performing optimally. The dataset includes the optimum twist values or torque values in the crankshaft 30 for a wide range of speed and other conditions. The dataset is stored in the form of a regression equation. The comparator 21 compares torque output or a quantity related to torque as measured and calculated in accordance with this invention with a standard or reference value in the dataset. The comparator 21, and control 22 mentioned below, in combination with the processor 14, form an engine management apparatus in accordance with the invention.

Based on said comparison, a signal is generated and sent to a control 22, forming part of the apparatus 10. The control 22 is configured to control an operating function of the engine e.g. ignition timing for a spark ignition engine, injector timing in the case of a diesel engine, boost pressure in the case of a turbocharged or supercharged engine, and the like, thereby to optimize engine performance. Thus it is possible to dynamically control engine performance at the level of individual pistons based on measurement of twist or torque or related quantity in the crankshaft 30 of the engine.

By having large numbers of datum points and measure points, the frequency of measurement during a revolution of the flywheel 40 is high and as a result the torque output of each piston 32 can be resolved accurately.

Advantageously, the processor 14 is preprogrammed during processing to process the measured data, for example by making use of Fast Fourier Transforms, to ameliorate the effect of vibrations and resonances in reciprocating and rotating components of the engine on the measured data and thus to isolate or to isolate to a degree, information which is useful in the context of this invention.

Further, advantageously, the engine management apparatus is preprogrammed, when the engine is operated under no load (for example when it

is idling), to calibrate the measuring apparatus in respect of datum time periods. This has, inter alia, the advantages that calibrating takes place regularly and thus also compensates for changes in ambient conditions and engine conditions (e.g. ambient and engine temperatures).

5 The Applicants believe that it is an advantage of the invention that it provides a system for monitoring engine performance which makes use of magnetic sensors which produce a well defined signal, are relatively robust, and have a good service life considering the harsh operating conditions of an engine. Further, by making use of the teeth of the flywheel and teeth on a disc fast with
10 the crankshaft as measure and datum points, the system can be manufactured cost effectively. Further, an advantage of the system is that it makes use of the time lag between the respective signals when the crankshaft 30 is under load relative to when it is under no load and is thus self-calibrating in that variations due to e.g. temperature effects, mechanical manufacture variations/tolerances etc.
15 are compensated for. The Applicants believe that by optimizing engine performance in accordance with the invention, the toxic content of exhaust fumes may be reduced. A further advantage of the invention is that it provides a system for real time measuring twist in a crankshaft during operation of an engine and to control operating functions in real time of the engine thereby to optimize engine
20 performance.

With reference to Figure 3, a gas turbine engine is generally indicated by reference numeral 129. The gas turbine engine is shown schematically, mainly in outline. The gas turbine engine 129 includes a main shaft 130 to which the current invention is applied. Conventionally, the gas turbine
25 engine includes a low pressure compressor represented by one stage 132.1 thereof adjacent an inlet of the gas turbine engine, a high pressure compressor downstream of the low pressure compressor, a combustion chamber, turbine stages 132.3 and an outlet.

In accordance with the invention, twist of the main shaft 130 is
30 measured and torque is derived from the measured twist. These measurements and calculations are effected in the same fashion as described with reference to Figures 1 and 2 for the reciprocating engine. Thus, also on the gas turbine engine, datum triggers 152 trigger a sensor 112.1 to generate a signal. The triggers 152

are conveniently provided on or by turbine vanes toward a rear end of the central shaft.

Furthermore, triggers are provided on or by the respective vanes of a first stage 132.1 of the low pressure compressor toward a front of the central shaft 130, and a sensor 112.2 is provided peripherally adjacent the pitch circle of the compressor vanes.

It is to be appreciated that stages other than the first stage of the low pressure compressor and the last stage of the turbine may be used to locate the triggers. Furthermore, it is to be appreciated that it is possible that a time lead as opposed to a time lag may be measured. For purposes of this invention a time lead is to be regarded as a negative time lag.

It is regarded as an important advantage that casings surrounding the compressor and turbine stages are of non-magnetic material, whereas the vanes are of magnetic material. Thus, the invention can be applied in a non-invasive manner, in so far as the sensors can be external of the casings.

It is to be appreciated that operation of a gas turbine engine approximates a steady state condition during any one revolution or small number of revolutions. Thus, first, torque fluctuations are expected to be substantially less than for a reciprocating engine. Thus, the Applicant envisages that a smaller number of triggers may be sufficient, also bearing in mind that, in gas turbine engines, very high speeds are sometimes used.

In other respects, performing the invention and the advantages obtained from the invention are substantially the same as explained with reference to Figures 1 and 2 in respect of the reciprocating engine.

CLAIMS

1. A method of measuring twist in a rotating shaft which is subjected to torque, between a datum point on the shaft and a measure point on the shaft longitudinally spaced from the datum point, the method including
- 5 establishing a datum time period between a datum time moment the datum point passes a stationary datum station and a time moment the measure point passes a stationary measure station under a no load condition and recording rotational speed of the shaft as the datum speed;
- 10 measuring a measure time period between a time moment the datum point passes the stationary datum station and a time moment the measure point passes the stationary measure station when the shaft is subjected to torque, recording the rotational speed of the shaft as the measure speed and establishing a length of the shaft over which torque is applied;
- 15 calculating the twist in the shaft on the basis of the difference between the measure time period and the datum time period, the measure speed and the length over which torque is applied.
2. A method as claimed in Claim 1 in which a torque point at which torque is applied to the shaft and a load point at which a load is connected to the shaft are longitudinally spaced, the shaft being stressed and undergoing twist
- 20 between the torque point and the load point, one of the measure point and the datum point being positioned along said stressed portion of the shaft, the other of the measure point and the datum point being positioned in a relaxed portion of the shaft beyond one of the torque point and the load point.
3. A method as claimed in Claim 2 in which the shaft is a crankshaft of
- 25 a reciprocating internal combustion engine.
4. A method as claimed in Claim 3 in which the reciprocating engine includes a ring gear having gear teeth at one end of the crankshaft and a disc at an opposed end of the crankshaft, the datum point being on the disc, the measure point being on a gear tooth of the ring gear.
- 30 5. A method as claimed in Claim 4 which is performed in respect of a

plurality of measure points and a plurality of datum points, the plurality of measure points being on a corresponding plurality of gear teeth of the ring gear and the plurality of datum points being points which are circumferentially spaced on the disc.

- 5 6. A method as claimed in Claim 5 in which the reciprocating internal combustion engine includes a plurality of cylinders, the method being performed in respect of each cylinder.
7. A method as claimed in Claim 5 or Claim 6 which is performed for each power stroke of the reciprocating internal combustion engine.
- 10 8. A method as claimed in Claim 2 in which the shaft is a main shaft of a gas turbine engine.
9. A method as claimed in Claim 8 in which at least one of the datum point and the measure point is on a vane of at least one of a compressor and a turbine of the gas turbine engine.
- 15 10. A method as claimed in Claim 9 which is preformed in respect of a plurality of measure points and a plurality of datum points, the plurality of measure points being on a corresponding plurality of vanes of one of the compressor and the turbine and the plurality of datum points being on a corresponding plurality of vanes of the other of the compressor and the turbine, the sensors being external
20 of casings surrounding respectively the compressor and the turbine.
11. A method as claimed in any one of Claim 1 to Claim 10 inclusive in which establishing the time moments includes triggering a sensor at respectively the datum station and the measure station by means of triggers at correspondingly the or each datum point and the or each measure point, creating a signal by each
25 sensor and recording the signal against time.
12. A method as claimed in Claim 11 in which the triggers are masses of magnetic material, and the sensors are responsive to said magnetic material to create said signals.

13. A method as claimed in Claim 11 in which the triggers are optically detectable surfaces, and the sensors sense passing of the triggers optically.

14. A method of measuring twist in a rotating shaft which is subjected to torque including sensing and recording a datum time moment when a datum
5 point on the shaft passes a fixed datum station; sensing, at a fixed measure station, an arrival time moment of a measure point on the shaft, longitudinally spaced from said datum point by a predetermined distance; measuring a measure time period between said datum time moment and said arrival time moment; measuring rotational speed of the shaft; comparing the measure time period with
10 a computed time period under a no load condition; and calculating the twist in the shaft on the basis of the time lag and the rotational speed.

15. A method as claimed in Claim 14 which includes empirically predetermining a datum time period between a datum time moment and an arrival time moment under a no load condition at a determined rational speed, and
15 computing a computed time period and a computed arrival moment for any specific rotational speed on the basis of the datum time period bearing in mind the respective rotational speeds.

16. A measuring apparatus for measuring twist in a rotating shaft which is subjected to torque, the measuring apparatus including
20 at least one datum trigger at a datum point on the shaft;
a datum sensor at a stationary datum station arranged to sense said at least one datum trigger when said at least one datum point is in register with the datum station and to generate correspondingly at least one datum signal;
at least one measure trigger at correspondingly at least one measure point
25 on the shaft longitudinally spaced from said at least one datum point;
a measure sensor at a stationary measure station arranged to sense said at least one measure trigger when said at least one measure trigger is in register with the measure station and to generate correspondingly at least one measure signal;
a clock keeping time;
30 a rotational speed meter for measuring and recording rotational speeds of the shaft against time respectively as the datum speed and as the measure speed;
recording means for recording said datum and measure signals against time;

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a processor programmed to establish

correspondingly at least one datum time period between a time moment said at least one datum point passes the datum station and a time moment said at least one measure point passes the measure station under no load
5 conditions and recording the rotational speed of the shaft as the datum speed,

correspondingly at least one measure time period between a time moment said at least one datum point passes the stationary datum station and a time moment said at least one measure point passes the stationary measure station when the shaft is subjected to torque and recording the rotational speed of the
10 shaft as the measure speed, and

the twist in the shaft on the basis of the difference between said at least one measure time period and said at least one datum time period, the measure speed and a length of the shaft over which torque is applied.

17. A measuring apparatus as claimed in Claim 16 in which the shaft
15 includes a torque point at which torque is applied to the shaft in use and a load point at which a load is connected to the shaft in use, the torque point and the load point being longitudinally spaced, the shaft, in use, being stressed and undergoing twist between the torque point and the load point, one of the or each measure point and the or each datum point being positioned along said stressed
20 portion of the shaft, the other of the or each measure point and the or each datum point being positioned in a relaxed portion of the shaft beyond one of the torque point and the load point.

18. A measuring apparatus as claimed in Claim 17 in which the shaft is a crankshaft of a reciprocating internal combustion engine.

25 19. A measuring apparatus as claimed in Claim 18 which includes a plurality of measure points and a plurality of datum points.

20. A measuring apparatus as claimed in Claim 19 in which the reciprocating engine includes a ring gear having gear teeth at one end of the crankshaft and a disc at an opposed end of the crankshaft, the datum points and
30 the measuring points being respectively on gear teeth of the ring gear and at circumferentially spaced positions on the disc.

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21. A measuring apparatus as claimed in Claim 17 in which the shaft is a main shaft of a gas turbine engine.
22. A measuring apparatus as claimed in Claim 21 in which at least one of the datum point and the measure point is on correspondingly at least one of a
5 vane of a compressor and a vane of a turbine of the gas turbine engine.
23. A measuring apparatus as claimed in Claim 22 which includes a plurality of datum points and a plurality of measure points, the datum points and the measure points being respectively on vanes of the compressor and the turbine.
24. A measuring apparatus as claimed in any one of Claim 16 to Claim
10 23 inclusive in which the triggers are masses of magnetic material, and the sensors are responsive to said magnetic material to generate signals.
25. A measuring apparatus as claimed in Claim 24 in which the sensors are in the form of magnetic sensors arranged external of casings surrounding respectively the compressor and the turbine.
- 15 26. A measuring apparatus as claimed in any one of Claim 16 to Claim 23 inclusive in which the triggers are optically detectable surfaces, and the sensors are optical sensors responsive to the optically detectable triggers to generate signals.
- 20 27. A method of measuring torque in a rotating shaft, including measuring twist in the rotating shaft in accordance with any one of Claim 1 to Claim 13 inclusive, and calculating torque on the basis of the measured twist bearing in mind a pre-established length of the shaft over which length said twist is measured, and pre-established physical characteristics of the shaft determining torque-twist behaviour of the shaft.
- 25 28. A method as claimed in Claim 27 in which said physical characteristics of the shaft determining torque-twist behaviour of the shaft are established empirically.

29. A measuring apparatus for measuring torque in a rotating shaft, which includes a measuring apparatus for measuring twist as claimed in any one of Claim 16 to Claim 26 inclusive, in which the processor is programmed to calculate torque on the basis of the measured twist bearing in mind a pre-established length of the shaft over which length said twist is measured and pre-established physical characteristics of the shaft determining torque-twist behaviour of the shaft.
30. A method of managing operation of an engine including measuring a quantity related to torque in a drive shaft of the engine as claimed in any one of Claim 1 to Claim 13 or Claim 27 or Claim 28, comparing the measured value of said quantity related to torque to a predetermined standard value of said quantity, establishing any deviation between the measured value and the standard value and controlling an operating function of the engine in response to said establish deviation.
31. A method as claimed in Claim 30 in which said operating function of the engine is at least one of ignition timing when the engine is a spark ignition engine, injector pump timing when said engine is a diesel engine, boost pressure when said engine is turbocharged or supercharged.
32. A method as claimed in Claim 30 or Claim 31 including automatically calibrating the measuring apparatus in respect of datum time periods when the engine is operated under a no load condition.
33. An engine including
- a measuring apparatus as claimed in any one of Claim 16 to Claim 26 or claim 29;
- an engine management apparatus including a comparator arranged to receive a signal indicative of a measured value of a quantity related to torque in a drive shaft of the engine from the measuring apparatus, the comparator being preprogrammed to compare said measured value with a standard value of said quantity related to torque to generate a control signal, the engine management apparatus being responsive to said control signal to control an operating function of the engine.

34. An engine as claimed in Claim 33 in which said operating function of the engine is at least one of ignition timing when the engine is a spark ignition engine, injector pump timing when said engine is a diesel engine, boost pressure when said engine is turbocharged or supercharged.
- 5 35. An engine as claimed in Claim 33 or Claim 34 in which the engine management apparatus is preprogrammed to calibrate the measuring apparatus in respect of datum time periods when the engine is operated under a no load condition.

ABSTRACT

In a reciprocating engine, power strokes of pistons apply torque to a crankshaft causing it to twist proportionally. Datum time moments are established by triggers rotating with, at one end of, the crank shaft and a sensor sensing passing of the triggers. At an opposed end, ring gear teeth trigger a sensor. At no load, zero twist occurs and arrival time moments of the triggers in relation to the datum time moments are predictable. Under load, twisting in the crank shaft causes the arrival time moments to be delayed, the time lag being proportional to the twist, bearing in mind the effect of rotational speed. With information about physical twist-torque characteristics of the crankshaft, torque values are calculated from the measured twist values, and are used in engine management. The invention can be applied to any shaft, for example also a main shaft of a gas turbine engine.

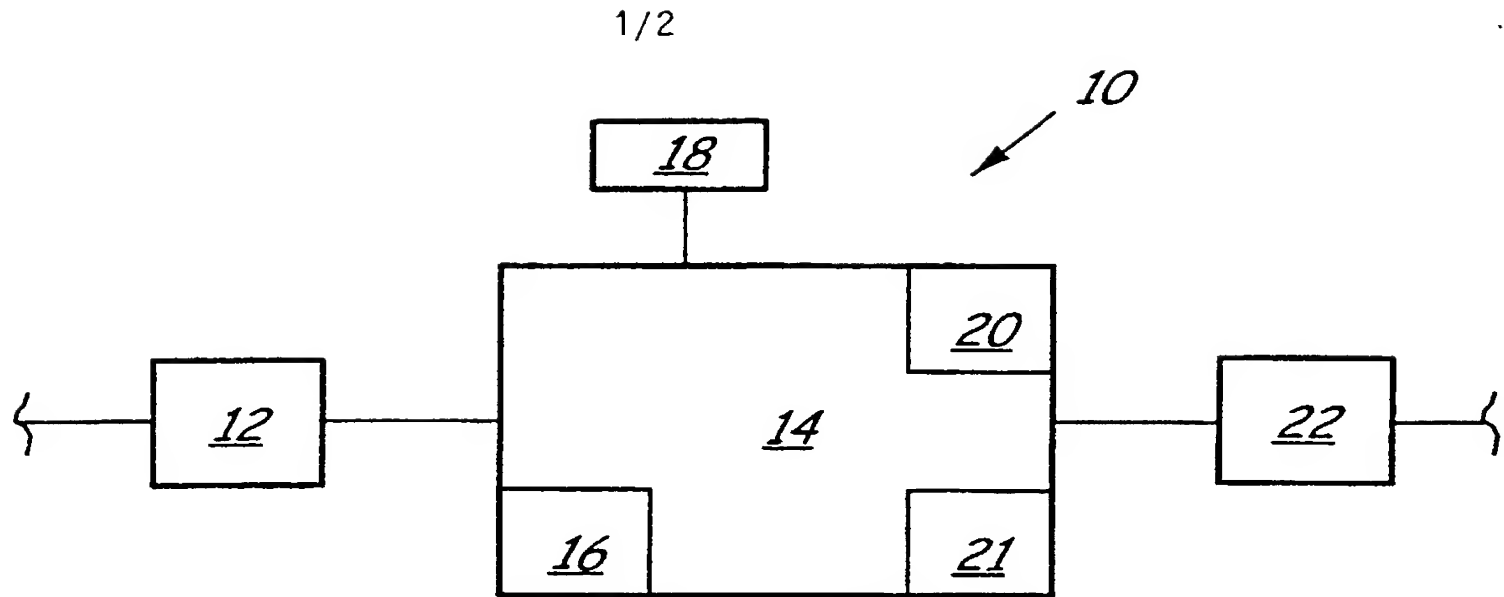


FIG 1

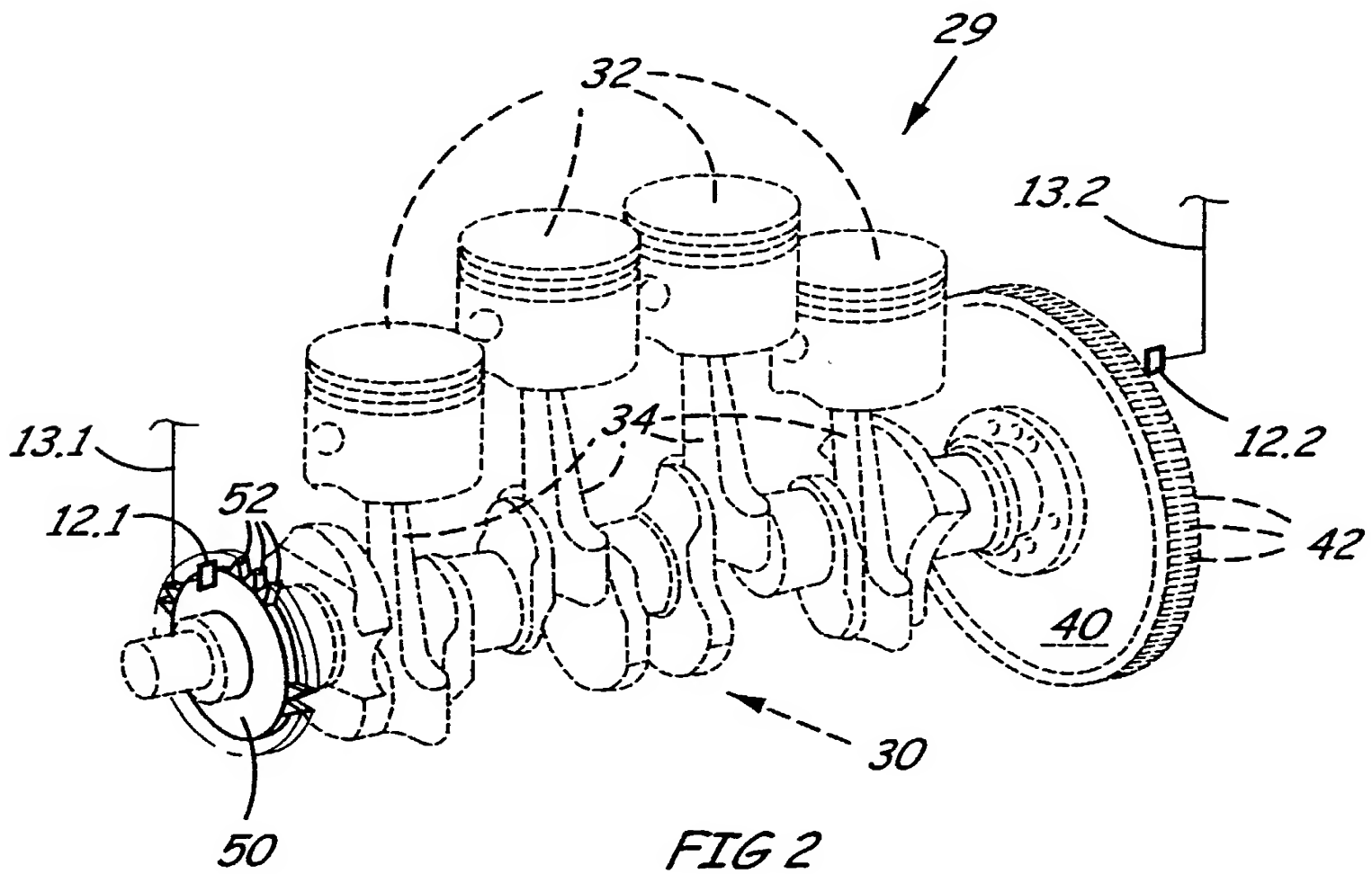
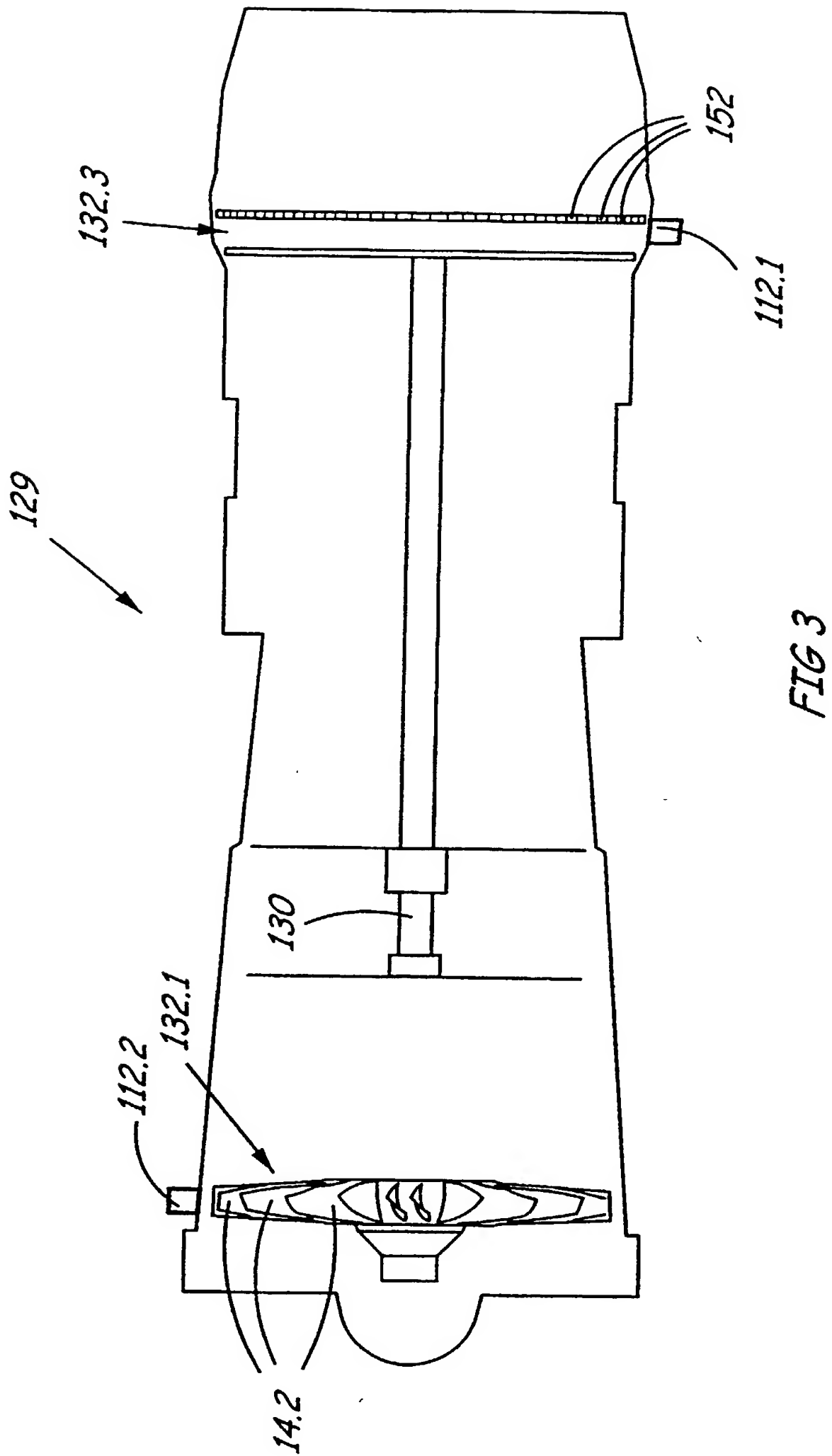


FIG 2

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DECLARATION FOR PATENT APPLICATION
AND POWER OF ATTORNEY (Under 37 CFR § 1.63)

FLH Docket No.

I hereby appoint William S. Frommer, Registration No. 25,506, or their duly appointed associate, my attorneys, with full power of substitution and revocation, to prosecute this application, to make alterations and amendments therein, to file continuation and divisional applications thereof, to receive the Patent, and to transact all business in the Patent and Trademark Office and in the Courts in connection therewith, and specify that all communications about the application are to be directed to the following correspondence address:

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I hereby declare that all statements made herein of my own knowledge are true and that all statements made on information and belief are believed to be true; and further that these statements were made with the knowledge that willful false statements and the like so made are punishable by fine or imprisonment, or both, under Section 1001 of Title 18 of the United States Code and that such willful false statements may jeopardize the validity of the application or any patent issued thereon.

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